Mars Trace Gas Mission Science Rationale & Concept

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Presentation to the NRC Decadal Survey
Mars Panel
10 September 2009

Background (1 of 2)

MEPAG Science Analysis Group Activities (2006-2007)

- Science Analysis Group (SAG-1), chaired by C.B. Farmer
 Strategic Mission to study atmospheric photochemistry and aeronomy, following up on Solar System Exploration Decadal Survey recommendation
 - ⇒ MAVEN and TGE selected to compete for 2011 launch as a Mars Aeronomy Scout
- Science Analysis Group (SAG-2), chaired by W. Calvin
 Defined 3 mission concepts, one of which was a Trace Gas mission to follow up on
 potential exchanges of methane between the atmosphere and subsurface, implying a
 dynamic Mars with the possibility of a biochemistry
 - ⇒ NASA forms Science Definition Team for a Mars Science Orbiter focused on trace gas detection and mapping

2013 MSO Science Definition Team (SDT)

- Telecons/meetings October–December 2007
- Final written report January 2008
- ⇒ Aeronomy Mars Scout (MAVEN or TGE) slipped to 2013 launch
- Report to MEPAG February 2008
- ⇒ MAVEN selected for launch in 2013

Background (2 of 2)

2016 MSO SDT Telecon Update:

- ⇒ Earth-based observations confirm methane detection, report variability
 - Focused on minimum payload required to follow up on reported methane discoveries, for possible low-cost NASA mission or joint ESA-NASA mission
 - Telecon held February 17, 2009
 - Briefing to NASA MEP and to MATT-3

ESA-NASA discussions on Joint 2016 Mission (2009)

Joint Instrument Definition Team (JIDT) studies Trace Gas instruments for ExoMars orbiter/carrier; re-affirms need for both detection of broad suite of gases in atmosphere and mapping of key trace gases

- ⇒ Earth-based observations confirm methane detection, report variability
- ⇒ ExoMars slips to 2018 opportunity

Study of joint mission combining Trace Gas orbiter with technology drop-off package is initiated (August, 2009)

Agenda

- Science Rationale & Objectives for a Trace Gas Mission
- A Mission Concept (NASA only) to Achieve those Objectives

What Science Questions are Raised by Methane Detection?

Current photochemical models cannot explain the presence of methane in the atmosphere of Mars and its reported rapid variations in space and time. Neither appearance nor disappearance can be explained, raising the following scientific questions:

- Is there ongoing subsurface activity?
 - Are there Surface/near-Surface Gas Reservoirs (particularly ice)? Where are they?
- What is the nature of gas origin: geochemical or biochemical?
 - Are other trace gases present? What are the isotopic ratios?
 - Nature of the methane source requires measurements of a suite of trace gases in order to characterize potential biochemical and geochemical processes at work
- What processes control the lifetimes of atmospheric gases?
 - Time scales of emplacement or activation and modification (seasonal, annual, episodic, longer term)
 - Role of heterogeneous chemistry: reactions on surface or airborne dust and ice
 - Atmospheric-surface exchange and atmospheric transport
- ⇒ What is the inventory, transport, and photochemistry of the Mars atmosphere? Note: It's not just about methane!

A Trace Gas Mission must provide:

- A comprehensive survey of both known gases (H₂O, H₂O₂, CO, etc.), as well as to improve detection limits by an order of magnitude or more on gases not yet observed.
- A *definitive statement* about whether or not methane is still present in the atmosphere and characterize whatever variability it has. However, a detailed inventory of atmospheric gases and their isotopologues would be a *major advance* in our understanding of the recent history and climate of Mars whether or not methane is detected.
- Characterization of the *roles that aerosols and temperature play* in controlling atmospheric composition.
- All atmospheric fields required (temperature, density/pressure, wind, aerosol concentration) to enhance our ability to *understand and to simulate for science and engineering* the Mars atmosphere.

Atmospheric Composition

Atmospheric evidence for present habitability

Key measurement objectives:

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Photochemistry (H<sub>2</sub>O<sub>2</sub>, O<sub>3</sub>, CO, H<sub>2</sub>O)
Transport (CO, H<sub>2</sub>O)
Isotopic Fractionation (isotopomers of H<sub>2</sub>O and CO<sub>2</sub>)
Surface exchange (CH<sub>4</sub> and H<sub>2</sub>O)
Inventory (HO<sub>2</sub>, NO<sub>2</sub>, N<sub>2</sub>O, C2H<sub>2</sub>, C<sub>2</sub>H<sub>4</sub>, C<sub>2</sub>H<sub>6</sub>, H<sub>2</sub>CO, HCN, H<sub>2</sub>S, OCS, SO<sub>2</sub>, HCI)
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Measurement goals:

Solar occultations to obtain sensitivity of 1–10 parts per trillion Limb-geometry mapping at sensitivity of 1–10 parts per billion with latitude/longitude/altitude/local time coverage

- ⇒ Would significantly improve knowledge of atmospheric composition and chemistry
- ⇒ Could lead to identification of source regions

Atmospheric State

Climate processes responsible for seasonal / interannual change

Key measurement objectives:

Wind velocity

Water vapor and atmospheric temperature without influence of dust

Diurnal coverage of all parameters

Vertical profiles of all parameters

Continue climatology monitoring

Measurement goals:

2-D wind velocity, temperature, aerosol optical depth, water vapor at 5 km vertical resolution over broad height range diurnal coverage twice per martian season 85% or better coverage along orbit

- ⇒ Extend record of climatology to characterize long-term trends
- ⇒ Validate and significantly improve models of transport and state

Surface Change Science

Recent processes of surface-atmosphere interaction

Key measurement objectives:

Geologic context of potential localized trace gas sources

Aeolian features (dust devil tracks, streaks, dust storm changes)

Gullies, avalanches, dune motions

Formation of small impact craters over time

Measurement goals:

1 meter resolution sufficient for these goals

- ⇒ Understanding active processes and the role of volatiles
- ⇒ Exchange of volatiles between high latitudes and atmosphere

Trace Gas Measurement Objectives

Detection:

- Would require very high sensitivity to the following molecules and their isotopomers: H₂O, HO₂, NO₂, N₂O, CH₄, C₂H₂, C₂H₄, C₂H₆, H₂CO, HCN, H₂S, OCS, SO₂, HCl, CO, O₃
- Detection sensitivities of 1-10 parts per trillion

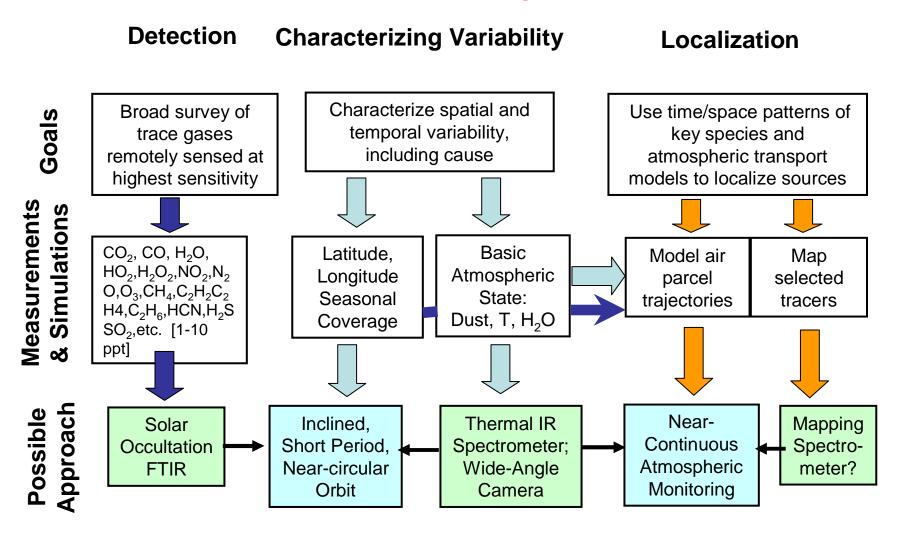
Characterization:

- Spatial and Temporal Variability: Latitude-longitude coverage multiple times in a Mars year to determine regional sources and seasonal variations (reported to be large, but still controversial with present understanding of Mars gas-phase photochemistry)
- Correlation of concentration observations with environmental parameters of temperature, dust and ice aerosols (potential sites for heterogeneous chemistry)

Localization:

- Mapping of multiple tracers (e.g., aerosols, water vapor, CO, CH₄) with different photochemical lifetimes and correlations would help constrain model simulations and points to source/sink regions
- To achieve the spatial resolution required to localize sources might require tracing molecules at the ~1 part per billion concentration
- Inverse modeling to link observed concentration patterns to regional transformations (e.g., in dusty air) and to localized sources would require simulations using circulation models constrained by dust and temperature observations

Trace Gas Measurement Requirement Flow-Down



Sample "strawman" Payload (existence proof)

Solar occultation spectrometer(s)

Atmospheric composition (broad spectral range and high resolution) Mapping key species (narrower spectral range)

Sub-millimeter spectrometer

Wind velocity through Doppler shift Water vapor, temperatures, etc., without influence from dust Map key species

Wide-angle camera (MARCI-like)

Daily global view of surface and atmospheric dust and clouds

Thermal-IR spectrometer

Daily global observations of temperature, dust, ice, water vapor Direct comparison to previous climatology record

High-resolution camera (as resources permit)*

Imaging of possible local sources and active surface processes *Prime difference between TGM concept and earlier MSO

Orbit characteristics:

Near-circular at low altitude (300-400 km)

- Would allow best coverage mapping
- Would allow most solar occultation opportunities (*most sensitive detection*)
- Orbit altitude might be increased at some point for planetary protection

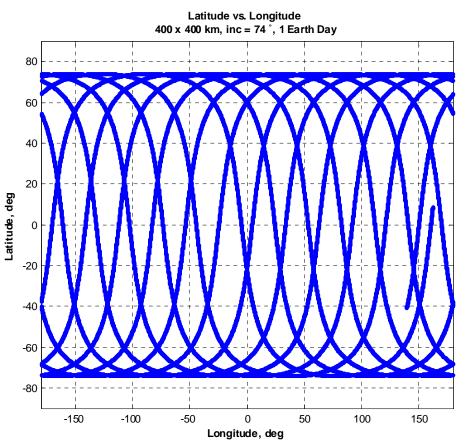
High inclination (~74° ±10°)

- Compromise between global coverage and faster precession of local time and more uniform latitude distribution of solar occultation points
- Science would require full diurnal cycle in less than a Martian season

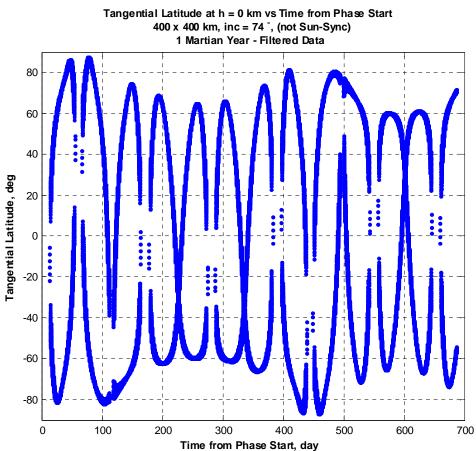
Mission duration (1 Mars year)

Full seasonal coverage

Orbit tracks for one day Good global mapping



Solar occultations for one year Good latitude distribution





OBJECTIVES

MISSION DESIGN

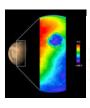
2016 TGM Mission Implementation Concept

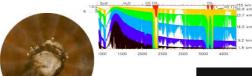
Perform Trace Gas / Mapping (TGM) Science

Atmospheric Composition and State Detection, Mapping, and Characterization

Telecom Infrastructure for future missions

Proximity UHF and deep space X band links







Nadir- and limb-pointing capability
UHF proximity telecom 250 Mb/sol
X-Band deep space telecom > 2 Gb/day
Data storage 64 Gb
EOL power 1500 W
Cost effective monopropellant propulsion
5 year lifetime, 10 years consumables

Notional Instruments*

Solar occultation (FTIR spectrometer)

Atmospheric composition

Wide-angle camera (MARCI-like)

Global view of surface, dust and clouds

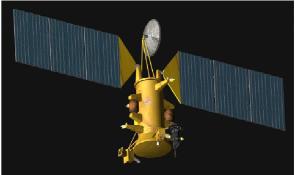
Thermal-IR spectrometer (TES-like)

Temperature, dust, ice, water vapor climatology

Mapping Spectrometer (Multiple Spectrum Measurement Approaches)

Water vapor, Wind, Temperature

Proposed Launch Date	January 2016
Launch C ₃	13.9 km ² /s ²
V_{inf}	3.8 km/s
Aerobraking Duration	6 to 9 months
Start Science Observation	March – June 2017*
Science Orbit	Walking, Inclination 74°
Science Emphasis Phase	~2 yrs, 400 x 400 km
Relay Emphasis	~2 yrs, 400 x 400 km



S/C Bus Dry Mass	1100 kg
Science Payload*	115 kg
Propellant	1915 kg
Total Wet Mass	3130 kg
LV Capability	3330 kg
(assuming an Atlas-V 411)	J

Notional Cost through Launch (RY \$M)*

Development Cost~ 535Launch Vehicle215Cost-though-Launch~ 750

Proposed Key Milestones

Mission Concept Review	~Sept 2010
PDR	Dec 2012
CDR	Sept 2013
Sys Integration Review	July 2014
Fight Readiness Review Dec 2015/Jan 2016	

*Several partnering approaches are being considered

Summary:

- TGM would enable *significant new science* and provide *key infrastructure* elements
- TGM science objectives not covered by any other proposed mission (including MSR)
- 2016 is favored launch opportunity for TGM:
 - Would provide needed telecom support for other future missions
 - Would minimize gap in atmospheric monitoring
 - Possible synergy with proposed MAVEN extended mission

Back-Up

MSO SDT Membership:

Michael Smith, Chair, NASA Goddard Space Flight Center Don Banfield, Cornell University Jeff Barnes, Oregon State University Phil Christensen, Arizona State University Todd Clancy, Space Science Institute Phil James, University of Toledo (retired) Jim Kasting, Pennsylvania State University Paul Wennberg, Caltech Daniel Winterhalter, JPL Michael Wolff, Space Science Institute Rich Zurek, JPL (Mars Program Office) Janis Chodas, JPL (MSO Project Manager) Tomas Komarek, JPL (MSO Mission Concept Manager)

JIDT Membership:

ESA Participants

- Augustin Chicarro
 - ESA Co-Chair
- Jean-Loup Bertaux
 - Service D'Aeronomie, CNRS
- Frank Daerden
 - BIRA/IASB
- Vittorio Formisano
 - IFSI Roma (I)
- Gerhard Neukum
 - Freie Universitaet Berlin
- Albert Haldeman
 - EXM/ESA

NASA Participants

- Richard Zurek
 - JPL (MPO) Co-Chair
- Mark Allen
 - JPL
- R.Todd Clancy
 - Space Science Institute
- Jim Garvin
 - NASA GSFC
- Michael Smith
 - NASA GSFC
- Tom Komarek
 - JPL (MPO)